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Thermal Instability

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Abstract: This paper discusses the thermal instability when a fluid is heated form below. The classic example of this, is a horizontal layer of fluid with its lower side hotter than its upper. The basic state is then one of rest with light fluid below heavy fluid. When the temperature difference across the layer is great enough, the stabilizing effects of viscosity and thermal conductivity are overcome by the destabilizing buoyancy, and an overturning instability ensues as thermal convection and an overturning instability ensues as thermal convection.

Keywords: Instability - Thermal conductivity-Benard convection - Raleigh number - Navier-Stockes equations

1. Introduction

Benard worked with very thin layers, only about 1 mm deep, standing on a levelled metallic plate which was maintained at a uniform temperature. The upper surface was usually free, and being in contact with the air was at a lower temperature. Various liquids were employed-some, indeed, which would be solids under ordinary conditions. The layer rapidly resolves itself into a number of *cells*, the motion being an ascension in the middle of a cell and a descension at the common boundary between a cell and its neighbours... . The cells acquire surfaces *nearly* identical, their forms being nearly regular convex polygons of, in general, 4 to 7 sides. The boundaries are vertical.... Fig.2.1 shows a plan of the convection cells in a silicone oil, with regular hexagons as the predominant polygons.

Stimulate by Bènard's experiments, Rayleigh (1916a) formulated the theory of convective instability of a layer of fluidon and boundary conditions to model the experiments, and derived the linear equation for normal modes. between horizontal planes. He chose equations of motion he then showed that instability would occur only when the adverse temperature gradient was so large that the dimensionless parameter $g \alpha \beta d^4 / \kappa v$ exceeded a certain critical value . Here g is the acceleration due to gravity, $\mathcal A$ the coefficient of thermal expansion of the fluid, b = -dQ/dz the magnitude of the vertical temperature gradient of the basic state of rest, d the depth of the layer of the fluid, κ its thermal diffusivity and η its kinematic viscosity. This parameter is now called the Raleigh number. We shall denote it by **Ra** in this paper, Ra. The Raleigh number is characteristic ratio of the destabilizing effect of buoyancy to the stabilizing effects of diffusion and

2. Thermal Instability

2.1Equations of Motion

dissipation.

2.1.1 The Exact Equations

The equations of motion of a heat-conducting viscous fluid under the action of gravity can be found in textbook (e.g. Batchelor 1967). In the notation of Cartesian tensors with position vector $\mathbf{x} = x_j$ and velocity $\mathbf{u} = u_j$ (j = 1,2,3), the equations are as follows. The equation of continuity is

$$\frac{\P r}{\P t} + \frac{\P (r u_j)}{\P x_j} = 0.$$
 (2.1.1)

The equations of motion are the Navier-Stockes equations,

$$\rho \frac{Du_i}{Dt} = -g \rho \delta_{i3} + \frac{\partial \sigma_{ij}}{\partial x_j}, (2.1.2)$$

where $D/Dt = \partial/\partial t + u.\nabla$, the x_3 -axis is the upward vertical, the stress tensor is given by

$$s_{ij} = -Pd_{ij} + m \begin{cases} \frac{3}{8} \|u_i\|_{L^2} + \frac{\|u_j\|_{L^2}}{\|x_i\|_{L^2}} + \frac{2}{3} \frac{u_k}{x_k} d_{ij} \frac{\ddot{\Theta}}{\ddot{\Theta}} \|u_i\|_{L^2} + \frac{u_k}{x_k} d_{ij},$$
(2.1.3)

 $m_{
m is}$ the coefficient of dynamic viscosity of the fluid, and

l is that of bulk viscosity (or secondviscosity). The equation of energy, or of heat conduction, is

$$r\frac{DE}{Dt} = \frac{\P}{\P x_j} \stackrel{\text{@}}{\not\in} \frac{\P q}{\P x_j} \stackrel{\overset{\overset{\circ}{\to}}{\to}}{\not=} \frac{P}{\P x_j} \frac{\P u_j}{x_j} + F, \quad (2.1.4)$$

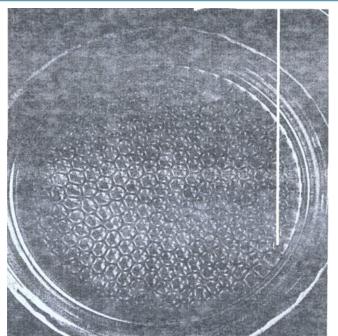


Figure 2.1.1: Be'nard cells under an air surface (from Koschmieder&pallas 1974)

where E is the internal energy per unit mass of the fluid, k is the thermal conductivity, q is the temperature, and the rate of viscous dissipation per unit volume of fluid is given by

$$F = \frac{1}{2} m \xi \frac{u_i}{\|x_j\|} + \frac{\|u_j \frac{\vec{0}}{\pm}}{\|x_i \frac{\vec{0}}{\pm}} + (l - \frac{2}{3}m) \xi \frac{u_k \frac{\vec{0}}{\pm}}{\|x_k \frac{\vec{0}}{\pm}}.$$

(2.1.5)

For a calorically perfect gas $E = c_v q$, and for a liquid E = c q, where c_n is the specific heat at constant volume and c the specific heat.

In general, the equations of state for a fluid specify ρ , μ , λ , k, c and Eas functions of P and q. For layers of real fluid in which the pressure does not vary much, these functions are almost independent of P.

The Boussinesq approximation

To these equations of motion, Rayleigh (1916a) applied the *Boussinesq approximation*, due independently to Oberbeck (1879) and Boussinesq(1903). The basis of this approximation is that there are flows in which the temperature varies little, and therefore the density varies little, yet in which the buoyancy drives the motion. Then the variation of density is neglected everywhere except in the buoyancy. On the basis of this approximation for small temperature difference between the bottom and top of the layer of fluid,

$$r = r_o \{1 - a (q - q_o)\},$$
 (2.2.1)

where r_o is the density of the fluid at the temperature q_o of the bottom of the layer and ${\cal A}$ is the constant

coefficient of cubical expansion. For a perfect gas, $a = 1/q_a \gg 3' \cdot 10^{-3} K^{-1}$, and for a typical liquid used in experiments $a \gg 5' \cdot 10^{-4} K^{-1}$. If $q_o - q = 10K$, then $(r - r_o)/r_o = a(q_o - q) = 1$, but nevertheless the buoyancy $g(\rho - \rho_0)$ is of the same order of magnitude as the inertia, acceleration or viscous stresses of the fluid and so negligible. For $d\mu/\mu d\theta$, $dk/kd\theta$, $dc/cd\theta$ α , so that m, k and c, or C_v , are treated as constants in the Boussinesq approximation. (The coefficient of bulk viscosity l is neglected, because it only arises as a factor of $\|u_j/\|x_j\|$, which is of order aalpha. In short, one approximates the thermodynamic variables as constants except for the pressure and temperature and except for the density when multiplied by g. This approximation works well for flows with temperature differences of a few degrees or less, such as those in Bènardexperiments, and can be formally justified by dimensional analysis (Spiegel &Veronis 1960, Mihaljan 1962). Here we shall give only a partial justification.

The differences of density in the continuity equation(2.2.1) are of order ${\mathcal A}$, so the approximation gives

$$\frac{\P u_j}{\P x_j} = 0,$$
(2.2.2)

as for an incompressible fluid. Then the stress tensor is given by

Again, on treating r and μ as constants in each term other than the buoyancy, the Navier-Stockes equations become

$$\frac{Du_i}{Dt} = -\frac{\partial}{\partial x_i} \left(\frac{p}{\rho_0} + gz \right) - \alpha g \left(\theta_0 - \theta \right) \delta_{i3} + v \Delta u_i,$$
(2.2.4)

where the Laplacian operator is given by $D = \P^2 / \P x_j^2$. Next we must simplify the heat equation (1.1.4). Firstly, note

that, if V is representative velocity scale of the flow, d a length scale, and Q_o – Q_1 a scale of temperature difference, then the ratio of the rate of production of heat by internal friction to the rate of transfer of heat is

$$F/r \frac{D(cq)}{Dt} \gg nV^2 d^{-2}/r_0 c(q_0 - q_1)V d^{-1} = nV/c(q_0 - q_1)d,$$

 q_1 being the temperature of the top of the layer of thickness d. Now, for a typical gas $n/c_v \gg 10^{-8} sK$ and for typical liquid $n/c \gg 10^{-9} sK$, which shows that the ratio is very

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small forboth gases and liquids unless $V/(q_0 - q_1)d$ is very large.

Therefore we shall neglect Φ . Secondly, note that the heating due to compression is

$$-P\frac{\P u_j}{\P x_i} = \frac{P}{r}\frac{Dr}{Dt} = aP\frac{Dq}{Dt}$$

For a perfect gas, $P = (c_p - c_v)r \ q$ and a = 1/qTherefore

$$r\frac{DE}{Dt} + P\frac{\P u_j}{\P x_i} @c_p r\frac{Dq}{Dt}, \qquad (2.2.5)$$

and the heating due to compression is not negligible in comparison to the heat transfer, as approximation (2.2.2) might have led one to expect. For liquids, however, the heating is negligible at normal pressure. The reason for this difference between gases and liquids is chiefly because, although the heating due to compression is typically only an order of magnitude less for a liquid than for a gas, the heat transfer is proportional to the density of the fluid, and a typical liquid is 10^3 times more dense than a typical gas. With all of these approximations, the heat equation becomes

$$\frac{Dq}{Dt} = kDq, (2.2.6)$$

where the thermal diffusivity $\kappa = k / \rho_0 c_p$ for perfect gas and $k / \rho_0 c$ for a liquid. Equations (2.2.3), (2.2.5) and (2.2.6) are called the *Boussinesq equations* and describe the motion of *Boussinesqfluid*.

3. The Stability Problem

3.1. The Linearized Equations

instability of Boussinesq fluid at rest between two infinite horizontal planes at different temperatures. Let the planes have equations $Z_* = 0$ and d, where the temperatures are q_0 and q_1 respectively. Here we denote a dimensional variable by subscripted asterisk to prepare for our choice of dimensionless variables; e.g. we shall soon take $z = z_* / d$ to be the dimensionless variable of height. Then the equations of motion give the basic state with

Rayleigh (1916a) modelled Bènard's experiments as the

$$U_*=0$$
, $Q_*=q_0$ - $bz_{*,(3.1.1)}$ $p_{*=}p_0-g\,
ho_0ig(z_*+rac{1}{2}lphaeta z_*^2ig)$ for $0\ \pounds\ z_*\ \pounds\ d$, where the basic temperature gradient $b=(q_0-q_1)d$. We anticipate that there can be instability only when $q_0>q_1$, i.e. when there is an adverse temperature gradient and b f 0 .

On putting

 $\mathbf{u}_* = \mathbf{u}'_*(\mathbf{x}_*, \mathbf{t}_*),$

 $q_* = \mathbf{Q}_*(z_*) + q_* \not (\mathbf{x}_*, t_*), P_* = P_*(z_*) + P_* \not (\mathbf{x}_*, t_*),$ and linearizing the Boussinesq equations for small

perturbations $\mathbf{u} \not \in q_* \not \in P_* \not \in \mathbf{r}$, it follows that

$$\tilde{N}_*.\mathbf{u} \not c = 0_{,(3.1.2)}$$

$$\frac{\P \mathbf{u} \mathbf{v}}{\P t_*} = -\frac{1}{r_0} \tilde{\mathbf{N}}_* P_* \mathbf{v} + a \mathbf{g} q_* \mathbf{k} + n \mathbf{D}_* \mathbf{u}_* \mathbf{v}_{, (3.1.3)}$$

$$\frac{\P q_{*}^{\not c}}{\P t_{*}} - b w_{*} \not c = k D_{*} q_{*}^{\not c}_{(3.1.4)}$$

In the absence of any basic velocity, we seek convection driven by buoyancy and moderated by viscosity and thermal diffusivity, so it is convenient to use scales d of length, d^2/k of time, and $bd=q_0-q_1$ of temperature difference. (One may equivalently use d^2/n as the time scale; this somewhat simpler if n? k.) Accordingly we

$$\mathbf{x} = \mathbf{x}_*/d$$
, $t = kt_*/d^2$, $\mathbf{u} = d\mathbf{u}'_*/\mathbf{k}$, (3.1.5)

$$q = q_* \not c / bd, P = d^2 P_* / r_0 k^2$$

Then the linearized stability equations (3. 1.2)- (3. 1.4) become

define the dimensionless variables

$$\tilde{N}.\mathbf{u} = 0_{.(3.1.6)}$$

$$\frac{\P \mathbf{u}}{\P t} = -\tilde{N}P + R \operatorname{Pr} q\mathbf{k} + \operatorname{Pr} \mathbf{D} \mathbf{u}, \quad (3.1.7)$$

$$\frac{\P q}{\P t}$$
 - w = Dq, (3.1.8)

respectively, where the dimensionless *Rayleigh number* is given by

$$R = gabd^4 / kn_{(3,1.9)}$$

and the Prandtl number by

$$Pr = n/k_{\cdot (3.1.10)}$$

Note that the Rayleigh number is positive when the lower boundary is the hotter one ($q_0 > q_1$) and is seen to be a characteristic ratio of the buoyancy to the viscous forces. Also note that the Prandtl number is an intrinsic property of the fluid, not of the flow; it measures the ratio of the rates of molecular diffusion of momentum and heat. We can now easily eliminate all the dependent variables except w, to get a single stability equation. The curl of equation (3. 1.7) gives

$$\frac{\P \omega}{\P t} = R \Pr(\tilde{N} q' \mathbf{k}) + \Pr D \omega, _{(3.1.11)}$$

where the velocity $\omega = \tilde{N}' u$. The curl of equation (3.1.11) in turn gives, after use of equation (2.1.6),

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$$\frac{\P}{\P t} \mathbf{D} \mathbf{u} = R \Pr(\mathbf{D} q \mathbf{k} - \tilde{\mathbf{N}} \frac{\P q}{\P z}) + \Pr \mathbf{D}^2 \mathbf{u} . (3. 1.12)$$

In particular

$$\frac{\P}{\P t} Dw = R Pr D_1 q + Pr D^2 w, (3.1.13)$$

where the horizontal Laplacian
$$D_1 = \P^2 / \P x^2 + \P^2 / \P y^2$$
. Finally elimination of

 θ from equations (3.1.8),(3.1.13) gives

$$\overset{\text{\tiny gen}}{\overset{\text{\tiny gen}}{\xi}} - D \overset{\overset{\text{\tiny od}}{\overset{\text{\tiny id}}{\overset{\text{\tiny od}}{\xi}}} 1}{\overset{\text{\tiny gen}}{\overset{\text{\tiny od}}{\overset{\text{\tiny od}}{\overset{\text{\tiny od}}{\xi}}}} P r \overset{\text{\tiny od}}{\overset{\text{\tiny od}}{\overset{\text{\tiny od}}{\overset{\text{\tiny od}}{\overset{\text{\tiny od}}{\overset{\text{\tiny od}}{\xi}}}} D w = R D_1 w. \ (3.\ 1.14)$$

Similarly it can be shown that heta satisfies the same equation.

It can be shown from the equation of continuity that

$$D_1 u = -\frac{\P^2 w}{\P x \P z} - \frac{\P w_3}{\P y}, \qquad (3.1.15)$$

$$D_1 v = -\frac{\P^2 w}{\P y \P z} + \frac{\P w_3}{\P x}, \qquad (3.1.16)$$

where $w_3 = \sqrt{y} - \sqrt{x} - \sqrt{y}$ is the vertical component of vorticity.

This is given by the vertical component of equation (3. 1.11), namely

$$\frac{\P w_3}{\P t} = \Pr D w_3 \qquad . (3.1.17)$$

So u, v can be found by solving the Poisson equations (3. 1.15), (3. 1.16) when W has been found by solving equation (3.1.14) and w_2 by solving the diffusion equation (3.1.17).

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