# Methods and Techniques for Detecting and Calculating the Trajectory of the Potential Asteroid *SNI2307*

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**Preface:** In June 2024, Author embarked on a journey of astronomical discovery by joining the Astronomy Club iAstronomer, run by Space India. Through a series of lectures and workshops, author developed a strong interest in celestial observation and learned about the All-India Asteroid Search Campaign- Phase-II, organized by the International Astronomical Search Collaboration (IASC), a NASA-affiliated initiative. From 1<sup>st</sup> to 26<sup>th</sup> July 2024, author participated in this campaign, having first attended a pre-campaign workshop that introduced me to the technical processes involved in asteroid detection. Equipped with the Astrometrica software and image datasets provided by the campaign, author encountered challenges in interpreting the initial black-and-white data. However, by delving deeper into the data's nomenclature and enhancing her understanding of celestial mechanics, author successfully identified a potential asteroid, which author named SNI2307. This discovery was formally recognized by the IASC and awarded author a Certificate of Appreciation for her contribution. This research paper provides an in-depth analysis of the image datasets and the steps leading to the discovery of SNI2307, offering a detailed look into the methods and tools employed in this research.

Abstract: During 1<sup>st</sup> to 26<sup>th</sup> July 2024, author participated in the All-India Asteroid Search Campaign- Phase-II<sup>[1]</sup>, organized by the International Astronomical Search Collaboration (IASC)<sup>[2]</sup> and Space India<sup>[3]</sup>, where author identified a potential asteroid, designated SNI2307. Using Astrometrica software<sup>[4]</sup> and data from a 1.8-meter Ritchey-Chrétien telescope<sup>[5,6]</sup>, author analyzed celestial images to differentiate moving objects from background stars. The object exhibited consistent motion, with shifts of 1.6 arcseconds in Right Ascension and 16 arcseconds in Declination, suggesting continuous movement. Magnitude variations from 21.3 to 20.9, likely due to changes in rotational orientation or surface reflectivity, along with a significant Signal-to-Noise Ratio (SNR)<sup>[7]</sup>, reinforced the object's classification as a potential asteroid. These results were submitted to the Minor Planet Center (MPC)<sup>[8]</sup> for verification. Author conducted further independent research to estimate its speed, momentum, size, and orbital trajectory, contributing valuable data to the study of Near-Earth Objects (NEOs)<sup>[9]</sup>. This research aims to enhance our understanding of celestial motion and improve detection methods for NEOs, ultimately supporting efforts in planetary defence. By integrating citizen science into this critical field, author hope to inspire future advancements in asteroid detection and promote wider engagement in astronomical research.

**Keywords:** SNI2307, Asteroid, Astrometrica, International Astronomical Search Collaboration, IASC, NASA, Space India, IAstronomer, Trajectory, Planetary defence, Near Earth Objects, NEO, Minor Planet Center, MPC, Pace Junior Science College

### 1. Introduction

Driven by a passion for planetary defence, author participated in the All-India Asteroid Search Campaign, an initiative organized by the International Astronomical Search Collaboration (IASC), a NASA<sup>[10]</sup> partner, in association with Space India. This campaign invites citizen scientists to engage in asteroid discovery and monitoring, leveraging community involvement to enhance global efforts in identifying Near-Earth Objects (NEOs). Since 2006, IASC participants have contributed substantially to asteroid research, with over **3,800** provisional detections and **120 numbered asteroids**<sup>[11]</sup>.

Near-Earth Objects (NEOs), which include asteroids and comets, are classified based on their orbits relative to Earth's distance from the Sun. Specifically, NEOs have a perihelion distance (the closest point to the Sun) of less than 1.3 astronomical units (AU). For context, 1 AU is the average distance from the Earth to the Sun, approximately 93 million miles or 150 million kilometres.

Near-Earth Asteroids (NEAs), a subset of NEOs, are categorized further into different groups based on their orbits:

- Atiras: Orbits entirely within Earth's orbit.
- Atens: Earth-crossing asteroids but having smaller semimajor axis.
- Apollos: Earth-crossing asteroids with larger semi-major axis.
- Amors: Earth-approaching asteroids orbiting in space between orbits of Earth and Mars.

This classification is crucial for understanding their potential risk to Earth, particularly as some NEOs are considered potentially hazardous due to their orbits bringing them close to our planet

Using Astrometrica software and data from a 1.8-meter Ritchey-Chrétien telescope, author analyzed celestial images to differentiate moving objects from background stars. One object exhibited consistent movement, suggesting continuous motion. Magnitude variations, along with a significant Signal-to-Noise Ratio (SNR), reinforced the object's classification as a potential asteroid. These results were submitted to the Minor Planet Center (MPC) for verification.

This research aims to analyze the object's real image data sets to estimate its speed, momentum, size, and orbital trajectory.

By contributing valuable data to the study of NEOs, this research hopes to enhance our understanding of celestial motion and improve detection methods for NEOs, ultimately supporting efforts in planetary defence. Additionally, this research underscores the value of citizen science in astronomy.

The following sections will delve into the analysis of the real image data sets using the methods outlined in the previous sections on asteroid detection methodologies. We will focus on:

- 1) **Optical Detection**: Analyzing the data to confirm the object's movement and potential asteroid classification.
- 2) **Signal-to-Noise Ratio** (SNR): Evaluating the data quality and determining the object's visibility.
- 3) **Data Cleaning and Noise Reduction**: Optimizing the data to improve the accuracy of trajectory analysis.
- 4) Application of Fourier Series and Calculus in Analyzing Object Trajectories and Motion: Utilizing mathematical tools to model the object's path.

# Research study on the Discovery of a Potential Asteroid "SNI2307" using Astrometrica Software:

In July 2024, author participated in the All-India Asteroid Search Contest, organized by the International Astronomical Search Collaboration (IASC), a NASA partner, and Space India. Through her involvement with the IAstronomica Club, author developed a deep passion for astronomy, fostered by numerous workshops and hands-on experiences with celestial mechanics. The campaign gave me the unique opportunity to translate this passion into scientific action, analyzing astronomical data to identify asteroids.

Using Astrometrica software and data from a 1.8-meter Ritchey-Chretien telescope (TEL F52) equipped with a CCD<sup>[12]</sup> camera, author worked meticulously over several weeks to analyze celestial images. The software enabled me to track object movement across the sky and differentiate between asteroids and background stars using Signal-To-Noise ratio (SNR) calculations.

Author identified a potential asteroid, preliminarily designated **SNI2307**, with precise positional data captured on July 2, 2024. **SNI2307**'s Right Ascension<sup>[13]</sup> (~20h 24m 09.25s) and Declination<sup>[13]</sup> (~  $-08^{\circ}$  18' 17.2"), alongside its magnitude of **21.3** as captured in the 1<sup>st</sup> data set, indicated it was faint but significant.

In recognition of her discovery, author was awarded a **Certificate of Appreciation** by the **International Astronomical Search Collaboration (IASC)** in **August 2024**. The discovery was a major milestone in her scientific journey, with **SNI2307** holding personal significance-its name reflects her initials (**SNI**) and birth date/year (**23/07**). Every time author gaze at the night sky, author feel a deep connection, knowing that her name is inscribed among the stars, forever linked to this asteroid.

### 1) Dataset Analysis for SNI2307:

SNI2307 was detected using Astrometrica software using four datasets, captured on July 2, 2024:

```
COD F52
OBS N. Primak, A.Schultz, S.Watters, J.Thiel, T.Goggia
MEA S. Keshari
TEL 1.8 - m f / 4.4 Ritchey - Chretien + CCD
ACK MPCReport file updated 9/17/2024 3:48:40 PM
NET PPMXL
     SNI2307
              C2024 07 02.52557 20 24 09.25 -08 18 17.2
                                                                   F52
                                                          21.3 G
              C2024 07 02.53880 20 24 08.72 -08 18 22.4
                                                          21.1 G
                                                                   F52
     SNI2307
              C2024 07 02.55197 20 24 08.18 -08 18 27.6
     SNI2307
                                                          20.9 G
                                                                   F52
              C2024 07 02.56518 20 24 07.65 -08 18 33.1 21.1 G
     SNI2307
                                                                   F52
```

----- end -----

# Output result of Astrometrica for the Celestial Object SNI2307

- a) C2024 07 02.52557 (20h 24m 09.25s, -08° 18' 17.2") -Magnitude: 21.3 G, F52
- b) C2024 07 02.53880 (20h 24m 08.72s, -08° 18' 22.4") -Magnitude: 21.1 G, F52
- c) C2024 07 02.55197 (20h 24m 08.18s, -08° 18' 27.6") -Magnitude: 20.9 G, F52
- d) C2024 07 02.56518 (20h 24m 07.65s, -08° 18' 33.1") -Magnitude: 21.1 G, F52

# 2) Inferring Data Sets:

a) COD F52

- **COD**: This refers to the Code for the observatory where the observation took place.
- **F52**: The Observatory Code assigned to this specific observation location, indicating which facility recorded the **SNI2307**'s position.

**Table I:** The image sets used in this research work have been captured by the following telescope

Observatory Code	Observatory Code Observatories, programs, surveys,		Region	MPC description	Latitude &
Map	and dedicated telescopes				Longitude
F52	Pan-STARRS 2 (PS2) at Haleakala	USA	Hawaii	Pan-STARRS 2,	19°36′N
	Observatory <sup>[6]</sup>			Haleakala	155°30′W

- b) OBS N. Primak, A. Schultz, S. Watters, J. Thiel, T. Goggia
- **OBS**: These are the Observers, the individuals who took part in collecting the data.
- N. Primak, A. Schultz, S. Watters, J. Thiel, T. Goggia: These are the names of the astronomers or citizen scientists involved in the observation.

#### c) MEA S. Keshari

- **MEA**: This refers to the Measurer—the individual responsible for processing and analyzing the data from the observation.
- S. Keshari: The name of the individual who measured SNI2307's position. That's me.

# d) TEL 1.8 - m f/4.4 Ritchey-Chretien + CCD

- **TEL**: This specifies the **Telescope** used for the observation.
- **1.8 m f/4.4 Ritchey-Chretien + CCD**: This describes a 1.8-meter Ritchey-Chretien telescope with an f/4.4 ratio (Focal Length to Aperture Diameter) equipped with a CCD (Charge-Coupled Device) camera, which is commonly used for precise astronomical imaging.

# e) ACK MPC Report file updated 9/17/2024 3:48:40 PM

- ACK: Acknowledgment indicating that this report was processed and acknowledged by the Minor Planet Center (MPC).
- MPC Report: A file containing observation reports submitted to the MPC.
- 9/17/2024 3:48:40 PM: The time and date the report was last updated.

# f) NET PPMXL

- **NET**: The reference **star catalogue** used for astrometric calculations.
- **PPMXL**<sup>[14]</sup>: PPMXL is a compilation of astrometric data for approx. 900 millions of stars used for astrometric purposes to determine the position of celestial objects.

### g) Observation Data Lines:

Each of the following lines provides specific information on **SNI2307**'s position over time:

• SNI2307: Designation of the observed celestial object (potential asteroid) defined by the MEA (myself) being tracked.

- C2024 07 02.52557: The date and time of the observation. Here, it's July 2, 2024, at 0.52557 part of the day (12:36:49.6 UTC).
- 20 24 09.25 -08 18 17.2: These are the Right Ascension (RA) and Declination (De), which specify the object's location in the sky:
- 20h 24m 09.25s RA (RA simulates the celestial equivalent of longitude).
- (-)08° 18' 17.2" De (De simulates the celestial equivalent of latitude).
- 21.3: The Apparent Magnitude ( $\hat{R}$ )<sup>[15]</sup> or brightness of the object (21.3).
- G: The G indicates the Green photometric filter used in the CCD observations.
- Each subsequent line records additional observations at different times to track the object's movement. For example:
- C2024 07 02.53880 corresponds to the next observation at 12:55:51.9 UTC with updated position coordinates.

# h) Magnitude Values:

The magnitude values (21.3, 21.1, 20.9, 21.1) reflect how bright the object appears. A higher number indicates a fainter object. For instance, 21.3 is quite faint, suggesting the object is not easily visible without a telescope.

- Analysis of Movement: SNI2307's position shifted approximately 1.604 arcseconds in Right Ascension (RA) and 15.780 arcseconds in Declination (De) across the datasets, indicating consistent motion. This change in position provides a strong signal that SNI2307 is a potential asteroid.
- **Reflectivity Changes: SNI2307**'s reflectivity magnitude fluctuated between **21.3** and **20.9**, likely due to its rotational orientation or surface properties. These brightness changes offer clues about its composition and movement as it reflects sunlight at different angles.
- Signal-To-Noise Ratio: Using Fourier Transformation and considering 1<sup>st</sup> frequency, the average Signal-To-Noise Noise Ratio is calculated as 5.345, which indicates that the data has a strong Signal as compared to its background Noise, indicating a high chance of being a potential Asteroid.
- Trajectory, Angular Velocity, Speed, and Size of SNI2307: Considering certain assumptions, the trajectory of the potential asteroid SNI2307 is as under:

Table II: Data on the research work to ascertain SN12507's trajectory.							
Field	Unit	Calculated Data	Table No.				
Average SNR Value	Ratio	5.345	Fourier Transformation				
Average Angular Speed	Arcseconds per Sec	0.008348068	Table IX				
Average Linear speed	Km/Sec	18.24566349	Table X				
Distance from Earth	AU	1.735948	Table XII				
Sun, Earth, and SNI2307	Angle in Degrees	152.7731942	Table XVI				
Distance from Sun	AU	2.664720186	Table XVII				
Specific Orbital Energy	Km <sup>2</sup> /sec <sup>2</sup>	-166.4326312	Calculated				
Eccentricity	Ratio	0.00005853908	Calculated				
Semi-Major Axis	AU	2.664876186	Calculated				
Semi-Minor Axis	AU	2.664876181	Calculated				
Specific Angular Momentum	AU <sup>2</sup> /second	0.00000325	Table XVIII				
Period of Revolution	Earth Years	4.353529302	Table XIX				
Estimated Size	%	Varies: 70.11% to 100% of max brightness	Table XX				
Avg Apparent Magnitude of SNI2307 (Ŕ)	Unitless	21.1	Table XXI				
Absolute Magnitude (H)	Unitless	17.77405838	Calculated				
Albedo (p) (assumed)	Unitless	0.075	Calculated				

#### **Table II:** Data on the research work to ascertain SNI2307's trajectory.

Estimated size of SNI2307 Km	1.35264139	Calculated	
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### i) Significance of the Data:

- **Positional Changes: SNI2307**'s consistent movement strongly indicates that it may be a potential asteroid, substantiated by mathematically calculated data for its orbit and trajectory.
- **Reflectivity Changes**: The fluctuations in brightness may point to rotational changes, revealing surface features that either reflect or absorb light differently.

### j) Analysis:

- This analysis indicates that the detected object is faint (magnitude 21.3), with a well-fitting PSF and a moderate signal-to-noise ratio (SNR = 6.0). The astrometric data (RA and De) pinpoints the object's location in the sky, while the PSF confirms it as a point source, ruling out an extended object like a galaxy.
- The designation "SNI2307" implies that the identified object is designated as SNI2307 by the MEA (myself), is under further observation by MPC.
- Using Astrometrica, a faint object has been identified, with real and measurable data. Further analysis and follow-up observations by MPC shall be required to understand the trajectory and behaviour of the discovered object.

# k) Significance of the Discovery:

This discovery represents a significant contribution to the growing catalogue of known asteroids. This campaign also highlighted the impact of **citizen science** in advancing our knowledge of **Near-Earth Objects (NEOs)** and improving planetary defence strategies.

# l) Further Observations and Analysis:

- SNI2307 is currently in Provisional Status, and with continued tracking over 6-10 years, it may eventually achieve Numbered Status. author may have the opportunity to propose a formal name for SNI2307 an honour for any budding astronomer.
- **Trajectory Calculation**: More precise observations and calculations by the **Minor Planet Center (MPC)** for a further period of 6-10 years will fully determine the nature of **SNI2307** and its trajectory. This research work aims to calculate the trajectory of the **SNI2307** based on the captured image data sets.

#### m) Surface and Rotation:

Variations in brightness suggest that **SNI2307** may have an irregular surface or rotate, presenting different reflective properties over time. The actual shape, size, and material content of the **SNI2307** shall be ascertained during further observations and analysis by MPC.

### 3) Challenges and Learning:

One of the key challenges author encountered during the campaign was refining techniques to clean noisy data and extract meaningful signals from interference. Author applied **Fourier Transformation** to analyze light frequencies and employed **vector algebra** and **calculus** to accurately determine asteroid trajectories. Understanding the significance of maintaining an **SNR** greater than 5 was crucial for identifying asteroids in the data sets. These challenges sharpened her data analysis skills and deepened her patience and persistence, essential traits for success in astrophysical research.

# 4) Analysis using Image Data Sets:



Image 1: With Data Analysis

Image 2: With Data Analysis



**Image 3:** With Data Analysis

#### 5) Key Data Points from Image 1:

#### a) **Object Detection**:

- The red circle indicates the detected celestial object. This is the candidate discovery of a potential Asteroid named **SNI2307**, an astronomical object under further investigation by MPC.
- The image data is being analyzed, and the software marks a potential detection with a specific designation.

#### b) Astronomical Datasets:

- Image files in FITS (Flexible Image Transport System) <sup>[16]</sup> format, containing stars and asteroid observations.
- c) **PSF (Point Spread Function) Fit:** 
  - The PSF Fit Parameters (x = 1155.27, y = 180.05) denote the object's location on the image's pixel grid, with x and y representing the horizontal and vertical pixel coordinates, respectively. These values are crucial for astrometric precision, as they define the exact point where the object is positioned within the image. This accuracy is essential for tracking the object's movement against reference stars. Precision is influenced by pixel resolution and telescope settings, with smaller deviations indicating greater accuracy.
  - SNR = 6.0: The Signal-to-Noise Ratio (SNR) of 6.0 indicates that SNI2307's signal is six times stronger than the background noise. In Astrometrica, this is calculated by comparing the object's brightness (signal) to the surrounding noise, with an SNR above 5 generally reliable for detection. Fourier series analysis can further refine this by separating periodic (signal) and random (noise) components. The formula for SNR is:

$$SNR = rac{S_{
m object}}{S_{
m background}}$$

- An SNR of 6.0 supports a credible detection of the asteroid, enabling precise tracking and measurement.
- Flux = 1020: This value represents the total light received from SNI2307, indicating its brightness. In asteroid detection, flux helps estimate the object's

Image 4: With Data Analysis

apparent magnitude, with a flux of 1020 suggesting that **SNI2307** is sufficiently bright for reliable detection.

- **FWHM = 0.8**": The Full Width at Half Maximum<sup>[17]</sup> (FWHM), expressed in arcseconds, indicates the width of an object's light profile at half its peak intensity. An FWHM of 0.8" (0.8 arcseconds) indicates minimal light spread due to atmospheric or instrumental factors, suggesting excellent resolution. This value signifies clear image quality, essential for accurate asteroid tracking, with typical values ranging from 1-2 arcseconds.
- Fit RMS = 0.155: The Fit RMS (Root Mean Square) of 0.155, measured in pixels, gauges the accuracy of the Point Spread Function (PSF) fit, which models how the asteroid's light is distributed in the image. The lower the RMS, the better the PSF model fit. An RMS below 0.2 is considered good, so 0.155 indicates a precise match, enhancing the accuracy of position and brightness measurements for reliable asteroid tracking.
- **Red curve overlay in scatter plot**: The red curve on the scatter plot represents the PSF (Point Spread Function) fit, which models the asteroid's light distribution. The white spots are actual observed data points, and their alignment with the red curve signals a good fit. A smooth, symmetrical bell shape of the red curve indicates a well-focused object, while deviations suggest data errors or noise. The bell curve height represents brightness, and a flatter red curve near the baseline indicates background noise.

#### 6) File Information:

- File Name: The FITS file being analyzed is named: "e60493h04450.762985.ch.696250.XY62.p10.fits."
- **Date and Time**: The observation was taken on July 2, 2024, at 12:36:49.6 UT.
- **RA** (**Right Ascension**) = 20h 24m 09.253s: This is the object's celestial longitude.
- **De** (**Declination**) =  $-18^{\circ}$  18' 17.20'': This is the object's celestial latitude.
- **R** = 21.3: The magnitude (brightness) of the object is 21.3, which is quite faint.

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#### 7) Object Designation:

The object has been designated by MEA as "SNI2307".

# 8) Position of SNI2307 and Sun at the time of Image capture (July 2, 2024):

Table III: Right Ascension (RA) of SIN12507								
Image Set	RA	Hours	Min	Sec	Hours	Degrees	Radians	
Image Set 1	20h 24m 09.253s	20	24	9.253	20.40257028	306.038554	5.341380	
Image Set 2	20h 24m 08.724s	20	24	8.724	20.40242333	306.036350	5.341342	
Image Set 3	20h 24m 08.185s	20	24	8.185	20.40227361	306.034104	5.341303	
Image Set 4	20h 24m 07.649s	20	24	7.649	20.40212472	306.031871	5.341264	
$\Delta RA =$	0.000445556	degrees arcseconds						
$\Delta RA =$	1.60400000							

Table III: Right Ascension (RA) of SNI2307

### Table IV: Declination (De) of SNI2307

Image Set	De	Degrees	Min	Sec	Degrees	Radians		
Image Set 1	-08 18' 17.27	-8	-18	-17.27	-8.304797222	-0.144946		
Image Set 2	-08 18' 22.39	-8	-18	-22.39	-8.306219444	-0.144971		
Image Set 3	-08 18' 27.64	-8	-18	-27.64	-8.307677778	-0.144996		
Image Set 4	-08 18' 33.05	-8	-18	-33.05	-8.309180556	-0.145023		
$\Delta De =$	0.004383333	degrees						
$\Delta$ De=	15.7799988	arcseconds						

The consistent shift  $\Delta RA$  (~1.6 arcseconds) and  $\Delta De$  (~16 arcseconds) over the four observations confirm that this object is moving relative to the background stars constellation-**PPMXL** (a compilation of astrometric data for approximately 900 million stars), a strong indication that it is a potential asteroid.

Additional information: The Right Ascension (RA) and Declination (De) of the Sun on July 2, 2024, can be determined using the Horizons system from NASA's Jet Propulsion Laboratory<sup>[18]</sup> (JPL), which provides precise celestial coordinates for solar system objects.

Table V: Right Ascension (RA) of the Sun at the time of Image Capture\*\*

Synchronous to Image Set	RA	Hours	Min	Sec	Hours	Degrees	Radians
Image Set 1	06h 48m 22s	6	48	22	6.806111111	102.091667	1.781836
Image Set 2	06h 48m 26s	6	48	29	6.808055556	102.120833	1.782345
Image Set 3	06h 48m 29s	6	48	29	6.808055556	102.120833	1.782345
Image Set 4	06h 48m 32s	6	48	32	6.808888889	102.133333	1.782563

Table VI: Declination (De) of the Sun at the time of Image Capture\*\*

Synchronous to Image Set	De	Degrees	Min	Sec	Degrees	Radians
Image Set 1	+22° 58' 09"	22	58	9	22.96916667	0.400888
Image Set 2	+22° 58' 05"	22	58	5	22.96805556	0.400868
Image Set 3	+22° 58' 01"	22	58	1	22.96694444	0.400849
Image Set 4	+22° 57' 57"	22	57	57	22.96583333	0.400829

\*\* The position of the Sun at the time of image capture<sup>[19a]</sup>. This places the Sun in the constellation **Cancer** during that time of the year.

# 2. Trajectory of SNI2307 under Black and White Background Image Data Sets



Image 1: Under Black Background

Image 2: Under Black Background



Image 3: Under Black Background

Image 4: Under Black Background



Image 1: Under White Background



**Image 2:** Under White Background



Image 3: Under White Background

# 3. Mathematical Calculations:

- 1) SNR Calculation using Fourier Transformation:
  - The flux data provided [1020, 1811, 2075, 1438] corresponds to signal intensities, while the SNR data ([6.0, 6.5, 5.5, 5.8]) gives a direct measure of signal-to-noise ratio for each set. To analyze the signal and background noise, the Discrete Fourier Transform (DFT) is applied to extract key frequency components.

Image 4: Under White Background

#### SNR=Flux/Noise

Flux data: [1020, 1811, 2075, 1438] SNR data: [6.0, 6.5, 5.5, 5.8]

#### a) Background Signal DFT

The calculated **Background Noise** for the data sets are: [170, 279, 377, 248]. The steps to calculate the DFT and extract amplitudes are as follows:

**Flux Data DFT:**  $X[k] = \sum_{n=0}^{N-1} x[n] e^{-i2\pi k n/N}$ 

Where: **x**[**n**] is the flux data for n image sets, **N=4** (number of points).

Fourier Transform (For simplicity,  $\mathbf{k} = \mathbf{0}$  and  $\mathbf{k} = \mathbf{1}$  are being considered as they are the dominant frequencies): For  $\mathbf{k} = \mathbf{0}$  (the DC component, with zero frequency): **X** [**0**] =1020+1811+2075+1438 **X** [**0**] =6344

# So, the Amplitude $A_0$ of the Signal = 6344/4 $A_0$ =1586.

For **k** = **1** (first frequency component): **X** [1] =1020+1811 x  $e^{-i2\pi/4}$ +2075 x  $e^{-i4\pi/4}$ +1438 x  $e^{-i6\pi/4}$ 

Simplifying using the Euler formula  $e^{i\theta} = \cos(\theta) + \text{author } x \sin(\theta)$ =>X [1] = (1020 + 1811 x (cos(- $\pi/2$ ) + author x sin(-

 $=>x [1] = (1020 + 1811 x (\cos(-\pi/2) + author x \sin(-\pi/2)) + 2075 x (\cos(-\pi) + ix \sin(-\pi)) + 1438 x (\cos(-3\pi/2) + author x \sin(-3\pi/2)))$ 

X [1] = -1055-373*i* 

The signal's amplitude  $A_1$  at this frequency is:  $A_1 = \{(-1055)^2 + (373)^2)\}^{(0.5)}$  $A_1 = 1118.9$ 

# b) Background Noise DFT

We apply the DFT formula:  $X[k] = \sum_{n=0}^{N-1} x[n]e^{-i2\pi kn/N}$ Where: **x[n]** is the flux data point for n image sets, **N=4** (number of points).

Fourier Transform (For simplicity, k=0 and k=1 are being considered as they are the dominant frequencies): For k = 0 (the DC component, with zero frequency): X [0] =170+279+377+248=1074

So, the **amplitude**  $A_0$  of the **Noise** =1074/4  $A_0$ =268.5

For  $\mathbf{k} = \mathbf{1}$  (first frequency component): **X** [1] =170+279  $\cdot e^{-i2\pi/4}$ +377  $\cdot e^{-i4\pi/4}$ +248  $\cdot e^{-i6\pi/4}$  Simplifying using Euler's formula  $e^{i\theta} = \cos(\theta) + author x \sin(\theta)$ :

 $=>X [1] = 170 + 279 x (\cos(-\pi/2) + \text{author } x \sin(-\pi/2)) + 377 x (\cos(-\pi) + \text{author } x \sin(-\pi)) + 248 x (\cos(-3\pi/2) + \text{author } x \sin(-3\pi/2))$ 

$$=>X[1] = -207 + 31i$$

The noise's amplitude  $A_1$  at this frequency is:  $A_1 = \{(-207)^2 + (31)^2)\}^{(0.5)}$  $A_1 = 209.3$ 

#### c) Calculation of final SNR Ratio inclusive of all Data Points after Fourier Transformation:

Using the amplitudes from 1<sup>st</sup> frequency (fundamental frequency) of the signal and noise from the DFT calculations:

- Signal Amplitude A<sub>signal</sub> = 1118.9
- Noise Amplitude  $A_{noise} = 209.3$

# The final SNR is calculated as:

- SNR= A<sub>signal</sub> / A<sub>noise</sub> = 1118.9/209.3 = **5.345**
- SNR value of 5.345 represents a good signal detection wrt noise, indicating that **SNI2307** may be a potential Asteroid.
- d) Calculation of Trajectory, Speed, and Size of SNI2307:

# Angular Distance Calculation:

Angle Between Two RA/De Points: Using the angular distance formula between successive RA and De positions, the change in angle (angular distance) between successive observations are calculated using the formula as under:

# $\begin{aligned} \cos \left( \Delta \theta_{12} \right) &= \operatorname{Sin}(\delta_1) \ x \ \operatorname{Sin}(\delta_2) + \operatorname{Cos}(\delta_1) \ x \ \operatorname{Cos}(\delta_2) \ x \ \operatorname{Cos}(\alpha_1 \\ &\quad - \alpha_2 \right)^{[19b]} \end{aligned}$

Where:

 $\delta_1$ ,  $\delta_2$  are Declinations calculated in between two image points, expressed in radians.  $\alpha_1$ ,  $\alpha_2$  are Right Ascensions calculated in between two image points, expressed in radians.

Table-VII: Calc	culating angular dis	tance traversed by	SNI2307:
Change in Angle	$\cos(\Delta \theta)$	$\Delta \theta$ (Radian)	$\Delta \theta$ (Degree)
I 0,100	0.00000000000	0 0000 45 4 4 4 ( 0 2 7	0.000(0070000

Change in Angle	$\cos(\Delta \theta)$	$\Delta \theta$ (Radian)	$\Delta \theta$ (Degree)
Between Image Set 1 & 2	0.99999999989674	0.0000454446037	0.0026037839913
Between Image Set 2 & 3	0.9999999989239	0.0000463917040	0.0026580488462
Between Image Set 3 & 4	0.99999999989122	0.0000466429887	0.0026724463949

# 2) Conversion of Observation Time to Seconds:

Table-VIII: Conversion of Observation Time to Seconds:

Image Time (Hr:Min:Sec)	Hours	Min	Sec	Time in Sec
Image 1: 12:36:49.6	12	36	49.6	45409.6
Image 2: 12:55:51.9	12	55	51.9	46551.9
Image 3: 13:14:50.3	13	14	50.3	47690.3
Image 4: 13:33:51.2	13	33	51.2	48831.2

# $\Delta T = T_4 - T_1 = 48831.2 - 45409.6$

 $\Delta T$ = 3421.6 seconds (Observation time Between Image 1 & 4)

#### 3) Angular Speed of SNI2307:

From calculus, the angular speed  $\omega$  can be calculated as  $\Delta\theta/\Delta t$  (change in angle of observations between each pair of observations and the time differences):

Table IX: Calculating Angular speed from Tables VII & VIII

Imaga Sata		Time between	$\omega$ (Angular Speed	$\omega$ (Angular Speed	$\omega$ (Angular Speed in
Image Sets	$\Delta \theta$ (Radian)	Images (Sec)	in Rad/Sec)	in Deg/Sec)	Arcseconds/ Sec)
Between Image Set 1 & 2	0.0000454446	1142.3	0.000000398	0.0000022794	0.008205920
Between Image Set 2 & 3	0.0000463917	1138.4	0.000000408	0.0000023349	0.008405636
Between Image Set 3 & 4	0.0000466430	1140.9	0.0000000409	0.0000023424	0.008432647

The last column of the calculation provides Angular Speed in Arcseconds per Second. The Average Angular Speed of SNI2307 calculated from the four image sets is  $8.348 \times 10^{-3}$  is Arcseconds per Second.

#### 4) Linear Speed of SNI2307:

Assuming that **SNI2307** is at a distance **1.735948** AU from Earth (1 AU =  $1.495978707 \times 10^8$  km (distance between Earth

and Sun)). This assumed distance of SNI2307 is derived from iteration (to ensure that SNI2307 distance is approximately between its semi-major and semi-minor axis of the ellipse). Using the angular speed and distance, the expected linear speed of SNI2307 is calculated as under:

**Linear Speed of SNI2307 (***V***) = Angular Speed** × **Distance** (1 AU = 149597870.7 Km)

1 CONTRACT

<b>Table X:</b> Calculating Linear Speed of SNI2307 (V)							
Image Sets	ω (Angular Speed	Linear Speed (V) in Km/Sec (Distance	Avg Linear Speed				
Image Sets	in Rad/Sec)	from point of observation = 1.735948 AU)	(V) in Km/Sec				
Between Image Set 1 & 2	0.00000039783	17.93498340					
Between Image Set 2 & 3	0.000000040752	18.37148548	18.24566349				
Between Image Set 3 & 4	0.000000040883	18.43052158					

Therefore, assuming that **SNI2307** is at **1.735948AU** distance away from the point of observation, **SNI2307** moving at an average speed of **18.24566349 Km/Sec**.

**Note:** The actual distance of **SNI2307** from the point of observation is still under study by MPC.

5) Calculate the Ordinates and Abscissa of the Sun and of SNI2307 with respect to Earth.

Table XI: Formula to calculate the Ordinates and Abscissa

of the celestial object with respect to the Earth						
P <sub>X</sub> co-ordinate	$P^*(Cos(\theta_{De})^*Cos(\theta_{RA}))$					
Py co-ordinate	$P^*(Cos(\theta_{De})^*Sin(\theta_{RA}))$					
Pz co-ordinate	$P^*(Sin(\theta_{De}))$					

Where  $\mathbf{P}$  is the distance magnitude between the celestial object under measurement with respect to Earth.

Using differential calculus, the speed of the celestial object between two image data points (a, b) can be calculated as:

# • Calculating the Sun's position vector with respect to the point of observation:

Distance  $R_{Sun}$  of Sun from Earth = 1 AU

1	<b>Table AII:</b> Calculating the Ordinates and Abscissa of the Sun with respect to					ш
	Image Set	R <sub>X(Sun)</sub>	R <sub>Y(Sun)</sub>	R <sub>Z(Sun)</sub>	Magnitude Rsun (AU)	
	Image Set 1	-0.192868013	0.900287744	0.390235709	1.0000000	
	Image Set 2	-0.193327873	0.900196845	0.390217853	1.0000000	
	Image Set 3	-0.193329462	0.900204244	0.390199998	1.0000000	
	Image Set 4	-0.193527442	0.900169442	0.390182143	1.0000000	

#### • **Calculating the SNI2307's position vector with respect to the point of observation:** Considering the Distance of **SNI2307** from Earth = 1.735948 AU (from Table X)

g the Distance of SIN12507 from Earth = 1.753946 AU (from Table X)

Table XIII: Calculating position vector of SNI2307 with respect to Earth:

Image Set	R <sub>X</sub> (Asteroid)	RY(Asteroid)	R <sub>Z</sub> (Asteroid)	Magnitude RAsteroid (AU)
Image Set 1	1.010599669	-1.389004637	-0.250738681	1.73594800
Image Set 2	1.010542571	-1.389038481	-0.250781320	1.73594800
Image Set 3	1.010484368	-1.389072928	-0.250825041	1.73594800
Image Set 4	1.010426353	-1.389106994	-0.250870094	1.73594800

1. Calculating the Speed of Sun and SNI2307 relative to the nearby Star:

The speed of the Celestial Object  $(V_{ab})$  for X, Y & Z directions is calculated as:

# Speed of the Celestial Object $(V_{ab}) = (P_b-P_a) / (T_b-T_a)$

#### Table XIV: Calculating Speed of the Sun relative to Alpha Centauri (nearby star)

Sun Speed Calculation (AU/sec)					Km/sec	Km/sec
Image Set	V <sub>x</sub> Component	Vy Component	Vz Component	$V_{ m Total}$	$V_{\mathrm{Total}}$	VAverage
Image Set 1 & 2	-0.0000004026	-0.000000796	-0.0000000156	0.0000004107	61.43386218	
Image Set 2 & 3	-0.0000000014	0.000000065	-0.0000000157	0.000000170	2.548387013	30.14790378
Image Set 3 & 4	-0.0000001735	-0.000000305	-0.0000000157	0.0000001769	26.46146216	

The Sun's actual speed relative to the nearby star Alpha Centauri (4.37 light-years away) is 19.4 km/s according to NASA's NSSDC<sup>[20]</sup>. In contrast, calculations based on three data sets yield a speed of 30.15 km/s. Increasing the number of data sets may lead to a more accurate determination of the Sun's speed.

	<b>Table XV:</b> Speed of SN12307 relative to PPMIXL (nearby constellation):					
	SNI2307 Speed Calculation (AU/Sec)					
Image Set	V <sub>x</sub> Component	Vy Component	Vz Component	$V_{\rm Total}$	$V_{ m Total}$	VAverage
Image Set 1 & 2	-0.000000868	-0.0000000514	-0.000000648	0.0000001199	17.93498395	
Image Set 2 & 3	-0.000000888	-0.000000525	-0.000000667	0.0000001228	18.37148464	18.24566359
Image Set 3 & 4	-0.000000883	-0.0000000518	-0.000000686	0.0000001232	18.43052221	

1 . C CNU2207 T-LL XX. C.

The value corroborates with the speed of SNI2307 calculated in Table X calculated from a different approach.

#### 2. Calculating the Angle $\theta$ between the Sun, the Earth and SNI2307

Dot Product = X(Sun) x X(Asteroid) + Y(Sun) x Y(Asteroid) + Z(Sun) x Z(Asteroid)

# $Cos(\theta) = \frac{Dot Product}{R(Sun)XR(Asteroid)}$

Table XVI: Angle Between vectors R<sub>(Sun)</sub> and R<sub>(Asteroid)</sub> using Table XII & XIII

Image Set	Dot Product	$\cos(\theta)$	$\theta$ (Radian)	$\theta$ (Degree)	$\theta$ (Degree) (Average)
Image Set 1	-1.543263388	-0.889003235	2.665960062	152.7482599	
Image Set 2	-1.543633452	-0.889216412	2.666425825	152.7749462	152.7731942
Image Set 3	-1.543677674	-0.889241886	2.666481511	152.7781367	152.7751942
Image Set 4	-1.543861926	-0.889348025	2.666713593	152.7914340	

#### Average Angle between Sun, Earth and SNI2307 is 152.77 degrees.



Geometric position of Sun, Earth and SNI2307 at the time of image capture.

3. Calculating the Distance between the Sun and SNI2307, assuming SNI2307 is at a distance of 1.73594800 AU from Earth:

Distance (Sun\_Earth): R<sub>Sun</sub> (1 AU) Distance (Earth\_Asteroid): RAsteroid (1.73594800 AU (From Table X)) Distance (Sun\_Asteroid): R (to be determined) The angle between  $R_{Sun}$  &  $R_{Asteroid}$  vectors:  $\theta$ = 152.77 Degrees.

**Using Distance formula:** 

 $\mathbf{R}_{(\text{Sun}_{\text{Asteroid}})} = [\mathbf{R}_{\text{Sun}^2} + \mathbf{R}_{\text{Asteroid}^2} - 2 \mathbf{x} \mathbf{R}_{\text{Sun}} \mathbf{x} \mathbf{R}_{\text{Asteroid}} \mathbf{x}$  $\cos(\theta)^{0.5}$ 

18	the <b>AVII</b> : Calculating the Distar	ice between the Sun ar	a SN12307:
Image Set	Distance R (Sun Asteroid) in AU	Average Distance R	Average Distance R
image Set	Distance R (Sun_Asteroid) III AU	(Sun Asteroid) in AU	(Sun Asteroid) in Km
Image Set 1	2.664590444		
Image Set 2	2.664729323	2.664720186	398636465.9
Image Set 3	2.664745918	2.004/20180	598050405.9
Image Set 4	2.664815061		

Colculating the Distance between the Sun and SNI2207

The estimated distance between the Sun and SNI2307 was approximately 2.664720186 AU (3.986 x 108 Km), based on the assumption made in Table X.

4. Calculating Specific Orbital Energy, Eccentricity, and Semi-major and Semi-minor Axis of revolution of SNI2307.

Using the formula for the Gravitational parameter of the Sun (G x M):  $\mu$  (km<sup>3</sup>/s<sup>2</sup>) = 1.327 x 10<sup>11</sup> km<sup>3</sup>/s<sup>2</sup> (G is Gravitational constant (6.67 x 10<sup>(-11)</sup>) and M is the mass of the Sun (1.989 x 10<sup>30</sup> Kg)). It is a key value used in orbital mechanics and celestial calculations.

# **Specific Orbital Energy (E):**

Specific Orbital Energy (E) is the energy per unit mass of an object in orbit around a central body, like a planet or star. It quantifies the total energy (kinetic and potential) associated with the object's motion, aiding in determining the orbit type (circular, elliptical, parabolic, or hyperbolic). The formula for Specific Orbital Energy is:

$$E=\frac{V^2}{2}-\frac{\mu}{R}$$

 $E (Km^2/sec^2) = (-)166.4326312$  (Negative indicates its orbit will form an ellipse)

• Calculating the Semi-major Axis of the Orbit of SNI2307:

Also,  $E = -\frac{\mu}{2a}$  where *a* is the Semi-major axis of the ellipse (orbit of **SNI2307**). Therefore, Semi-major Axis  $a = -\frac{\mu}{2E}$ .

 $a = 39\overline{8659803.088109}$  Km (2.664876186 AU)

Calculating Eccentricity e of the Elliptical Orbit of **SNI2307**:

$$e = 1 - \frac{R}{a}$$

R is the distance between Sun and **SNI2307** and *a* is the Semi-major Axis.

 $\mathbf{e} = 0.00005853908$  (~0, indicating that the orbit of SNI2307 is almost circular).

• Calculating the Semi-minor Axis of the Orbit of SNI2307:

Using the formula for the Eccentricity of an Ellipse:  $\mathbf{e} = (1 - \mathbf{b}^2 / \mathbf{a}^2)^{0.5}$ 

# The b (Semi-minor Axis) of the Orbit of SNI2307 = 398659802.405041 Km (2.664876181 AU).

The calculation of the assumed distance in **Table VIII** was to ensure that the calculated distance R in **Table XVII** (2.664720186 AU) is close enough to be within the limits of Semi-major and Semi-minor Axis. However, a minor variation of the distance is still possible as the object moves through various image data sets.

**Calculating the Specific Angular Momentum of SNI2307:** As **SNI2307** moves in an almost circular orbit, its velocity vector shall be tangential to its radial vector. Hence, its Specific Angular Momentum is calculated as:

#### Specific Angular Momentum (L): (R x V)

Using data from **Tables XIII and XV**, the Specific Angular Momentum for **SNI2307** is calculated as under:

Specific Angular Momentum (L)	R x V
In m <sup>2</sup> /second	7.27339 x 10 <sup>15</sup>
In Km <sup>2</sup> /second	7273386810
In AU <sup>2</sup> /second	3.25002 x 10 <sup>-7</sup>

• Calculating the Time Period of Revolution of SNI2307: Using the formula for the orbital period from Kepler's Third Law, the Time Period (T) of the Revolution is calculated as  $T^2=(2\pi)^2a^3/\mu$ .

Table XIX: Time Period of the Revolution of SNI2307:T (Seconds)137292900.1

4.353529302

Therefore, **SNI2307** is expected to revolve around the Sun in **4.3535 Earth Years.** 

#### 1. Determining Size of SNI2307:

T (Earth Years)

Using the flux data, the relative size of **SNI2307** at various positions can be identified. However, the exact size cannot be calculated using Astrometrica software.

The **flux data** captured in the 4 Image Data sets are: **[1020, 1811, 2075, 1438]**.

Average Flux captured in the 4 Image data sets = (1020+1811+2075+1438)/4 = 1586.

# Calculating the Relative Size of SNI2307 in different image data sets:

**Flux**  $\propto$  Projected cross-sectional area of **SNI2307** ( $\pi$ D<sup>2</sup>/4) where D is the projected diameter of **SNI2307** (considering it to be spherical in shape) facing the earth at the point of observation.

=> Relative size: Relative Size ∝ (Flux)<sup>(0.5)</sup>

Image Set	Flux	Flux <sup>1/2</sup>	Relative Size (relative Diameter) with respect to the maximum observed size	% of the Relative Size Observed	
Image Set 1	1020	31.94	70.11%		
Image Set 2	1811	42.56	93.42%		
Image Set 3	2075	45.55	100.00%	Maximum Relative Size Observed	
Image Set 4	1438	37.92	83.25%		

#### **Table XX:** Relative Size of SNI2307:

The above data signifies that **SNI2307** was at its maximum size in the  $3^{rd}$  observation set among the recorded 4 observations image sets, with the relative size in the  $1^{st}$  image set being 70.11% of the  $3^{rd}$  Image Set. This signifies that **SNI2307** is in continuous rotation with respect to the Earth,

resulting in different exposure surface areas at the different observation points.

• Average Apparent Magnitude  $(\hat{R})$ : The Apparent magnitude of SNI2307 can be estimated by averaging the magnitudes in the four-observation image data sets:

Table XXI: Apparent Magnitude "Ŕ" of SNI2307					
Image Set	Apparent Magnitude " <b>Ŕ</b> " of <b>SNI2307</b> from the	Average Apparent Magnitude " <b>Ŕ</b> " of <b>SNI2307</b> from			
illiage Set	given set of Data	the given set of Data			
Image Set 1	21.3				
Image Set 2	21.1	21.1			
Image Set 3	20.9	21.1			
Image Set 4	21.1				

Average Apparent Magnitude of **SNI2307** (**Ŕ**)= (21.3+21.1+20.9+21.1)/4= **21.1** 

• Limitations of Usage of Astrometrica Software Astrometrica primarily provides positional and flux

(brightness) data from astrometric measurements of celestial objects like asteroids. While it tracks the position, motion, and apparent magnitude of an asteroid, it does not directly output Albedo (p) or Absolute Magnitude (H), which are essential for calculating the asteroid's absolute size. Albedo measures reflectivity, indicating how much sunlight or radiation a celestial body reflects compared to what it absorbs. These data can be sourced elsewhere, beyond this research's scope. However, SNI2307's absolute size can be estimated under certain assumptions.

**Estimating Absolute Magnitude H from Astrometrica** Data

Calculate Absolute Magnitude H<sup>[21]</sup>: The absolute magnitude H is the apparent magnitude the asteroid would have if it were located 1 AU from both the Earth and the Sun. H can be calculated as per the following formula:  $H = \dot{R} - 5 \times \log_{10} (R_{Asteroid} \times R)$ 

where:

 $\dot{\mathbf{R}}$  is the Apparent Magnitude from Astrometrica.

RAsteroid is the distance of Asteroid from Earth in AU.

R is the distance of Asteroid from the Sun in AU.

Based on our assumption, the value of H is calculated for SNI2307 as:

H=21.1 - 5 x log<sub>10</sub>(1.735948 x 2.664720) H=17.77405838.

This gives you the absolute magnitude H, which is essential for estimating the size of SNI2307.

### Estimating Albedo (p)

- a) Albedo Data:
- Astrometrica does not provide albedo information. Albedo values must be sourced from other observations, such as infrared data from space-based missions (e.g., NASA's NEOWISE<sup>[22]</sup>) or using a standard assumption based on the asteroid's taxonomic type.
- Typical albedo values range from 0.05 (dark, carbonaceous asteroids) to 0.3 (bright, rocky asteroids).

#### b) Typical Albedo data:

We can use an estimated value based on the asteroid's classification. Typical values for Albedo for different types of Asteroids<sup>[23]</sup> are as under:

- C-type (Carbonaceous): Representing over 75% of known asteroids, C-types are very dark with an albedo of 0.05-0.10. Their composition is similar to the Sun but lacks hydrogen and volatiles. They mainly occupy the outer main belt.
- S-type (Silicaceous): Making up about 17%, S-types are brighter (albedo 0.15-0.25) and consist of iron and magnesium silicates. They dominate the inner belt.
- M-type (Metallic): Found mostly in the middle belt, Mtypes have an albedo of 0.10-0.30 and are rich in metallic iron.

Assuming SNI2307 is a C-type Asteroid (which is most abundantly available) with a typical Albedo value of 0.075, the size of SNI2307 is calculated as per the below formula:

D=1329/(p)<sup>(0.5)</sup> x 10<sup>-0.2H</sup>  $D = 1329 \bar{/} (0.075)^{(0.5)} \times 10^{-0.2 \times 17.77405838}$ 

#### D = 1.35264139 Km.

The estimated size of SNI2307 based on the above assumptions is 1.3526 Km<sup>[24]</sup>.

# Conclusion

In conclusion, the journey of discovering SNI2307 has been both a personal and scientific milestone, highlighting the intersection of passion and purpose in the realm of astrophysics and planetary defence. Though SNI2307 is currently positioned 1.73 AU from Earth, our ongoing analysis of its trajectory will be crucial in determining its future classification as a Near-Earth Object (NEO). Specifically, if its trajectory brings it closer than 0.3 AU from our planet, it could represent a potential threat that necessitates further monitoring and study.

This research not only contributes to the growing body of knowledge surrounding asteroid detection but also underscores the importance of collaborative efforts in planetary defence. By submitting our findings to the Minor Planet Centre, we have taken an essential step toward validating SNI2307's status and understanding its long-term orbital behaviour. The process of confirming its orbit, which may take 6 to 10 years, is critical in assessing the asteroid's potential impact risks.

This research work on the celestial object SNI2307 brings a key understanding on the motion of the potential asteroid, the cosmos' vast potential, and our duty as explorers. As author continue her journey in astrophysics, author remain committed to advancing our understanding of asteroids and enhancing our capabilities in safeguarding Earth from potential threats. Through research and vigilance, we can ensure our preparedness to face the challenges that may arise from these celestial bodies.

# Acknowledgment

I would like to express her sincere gratitude to the International Astronomical Search Collaboration (IASC) for providing the invaluable opportunity to participate in the All-India Asteroid Search Campaign. her heartfelt thanks also go to Space India and the IAstronomer club for their support and resources, which have significantly enriched her astronomical knowledge and skills. The datasets from the Astrometrica software were instrumental in enabling the discovery of the potential asteroid, and author am deeply appreciative of this powerful tool and thank the **Observers** of these Image data sets.

I extend her profound appreciation to her principal, Mr. Rahul Yadav of Pace Junior Science College, Powai, for his unwavering support and guidance throughout her journey in classes 11 and 12. author would also like to acknowledge Mr. Swapnil Kumbhar, her Physics teacher, for his insightful evaluations and guidance, which helped me structure this research paper effectively.

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calculating the observational data sets, and assisting in writing this research paper has been invaluable.

This discovery would not have been possible without the collective support and encouragement of all those mentioned above.

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# **Author Profile**



**Snigdha Keshari**, aged 16, is currently a Class 12 Science student at Pace Junior Science College, Powai, Mumbai, India. She has a keen interest in astrophysics, with two published articles to her credit: *"Black Holes: The Key to Understanding Our Entire Universe"* and

"The Journey of Gravitational Waves: From Newton to Modern Detection." In July 2024, during the All-India Asteroid Search Campaign, organized by the International Astronomical Search Collaboration (IASC), a NASA partner, Snigdha discovered a potential asteroid and designated it as SNI2307. For her significant contribution, she was awarded a certificate of appreciation by IASC. This current research paper, titled "Methods and Techniques for Detecting and Calculating the Trajectory of the Potential Asteroid SNI2307," is a continuation of her efforts to further the understanding of asteroid detection. Snigdha is passionate about pursuing a career in astrophysics and contributing to the field through scientific research and exploration.

A short poem on her Discovery of the Asteroid "I found a planetoid of a million worth, Miles away from the crust of Earth. I am up in the sky with feet on ground, In the vastness of space, her purpose is found."

Snigdha Keshari



Certificate of Appreciation from International Astronomical Search Collaboration Volume 13 Issue 10, October 2024 Fully Refereed | Open Access | Double Blind Peer Reviewed Journal www.ijsr.net



All India Asteroid Search Campaign: 2nd July'24-26 July'24.

From: AIASC Space <aiasc@space-india.com>
Date: Sat, 22 Jun 2024 at 22:19
Subject: AIASC 2024 Phase 2 Training workshop details
To: AIASC Space <aiasc@space-india.com>
Dear Participant,
Congratulations! We are happy to share that you have been selected for the prestigious All India Asteroid Search Campaign 2024, Phase 2.
SPACE India is the coordinator for the All India Asteroid Search Campaign in India. Under this highly recognized campaign, participants get exclusive access to images of the sky taken at observatories and analyze the data with specialized software provided during training to look for Asteroids. Objects reported by students could be potential discoveries. All observations contribute to the Near-Earth Object (NEO) data compiled by NASA and Jet Propulsion Lab (JPL).
AIASC Phase II will run from July 1st - 26th july 2024.

Email communication from Space India on selection for the Campaign

AIASC 2024 Phase 2 list All the communication with the participants will be done on the following mentioned email IDs.										
S. no	Team name	Participant 1	Email of 1st participant	Participant 2	Email Id of 2nd participant					
1	SPACE Titan	Yaser Soleimani	yaser.soleimani@gmail.com	Zahra Atri	rana.atri90@gmail.com					
2	SPACE Gemini Brothers	Farid Mehrabadi	magentaflamelake@gmail.com	Matin Bayati	bayatimatin4@gmail.com					
3	SPACE Andromeda	Sarina Mohammadi	sarinamohammadi971@gmail.com	Tannaz Mohajeri Parizi	Lajibolala172737@gmail.com					
4	SPACE Amazing Follower	Narges Payami	zhra83859@gmail.com	Zahra Darrudi	darrudizahra5@gmail.com					
5	SPACE Myrtilus	Melika Masihabadi	masihmelika86@gmail.com	Hadis Zargerani	Hadiszargerani1385@gmail.com					
6	SPACE Scorpius	Ayda Rashidabadi	aydarashidi1999@gmail.com	Amir Hosein Asaadi	asaadi.amir@gmail.com					
7	SPACE team 95	Swetha M	shwetham662@gmail.com	Kundan Raju Varhade	kundanvarhade2008@gmail.com					
8	SPACE team 96	S. Satya Bala Anusha	anushasatyavolu98@gmail.com	Sourabh Chandrakar	sourabhchandrakar88@gmail.com					
9	SPACE team 97	Lakshmi H R	lakshmihr2007@gmail.com	Avinnessh Srikanes	asrikanes26@gmail.com					
10	SPACE team 98	Akshita Gotmare	akshitagotmare@gmail.com	Humayu Rasheed	rasheedhumayu202@gmail.com					
11	SPACE team 99	Srilakshmi N	shrilak@gmail.com	Naman Jain	naman03jain@gmail.com					
12	SPACE team 100	Bhavya Jain	bjain1349@gmail.com	Naveen Negi	nnegi3939@gmail.com					
13	SPACE team 101	Devyani	choudharydevyani5@gmail.com	P Tanishka	neetu.1007@gmail.com					
14	SPACE team 102	Danny Damiaio Mantero	dannydmantero@gmail.com	Sharshti Mittal	shrashtimittal323@gmail.com					
15	SPACE team 103	Harishankar Nath Tiwari	harishankarnath1@gmail.com	Arnav Gupta	arnavgupta2606@gmail.com					
16	SPACE team 104	Abdul Rehman	khanchetayyaba2009@gmail.com							
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2	Space iAstronomer 16	Advaith R	vet04rahul@gmail.com	Vanya Agrawal	a Agrawal agrawalva@uws.edu.in					
3	Space iAstronomer 17	P Tanishka	neetu.1007@gmail.com	Yasaswi VSNS	callraj2002@hotmail.com					
4	Space iAstronomer 18	Shivam Dan	shivamdan24052003@gmail.com	Aarush Chinchole	chincholeaarush@gmail.com					
5	Space iAstronomer 19	Paridhi Ashish Shirke	smshirke01@gmail.com	Akshaj Mohit Vyas	mohitvyas2006@gmail.com					
6	Space iAstronomer 20	Myraa Khattar	Bhawnamyraa@gmail.com							
7	Space iAstronomer 21	Snigdha Keshari	snigdhakeshari123@gmail.com	Yadnya Sagar	mohitcbic@gmail.com					
8	Space iAstronomer 22	Pulkit Mahajan	ekta16mahajan@gmail.com	Sukhman Singh	Sukhman.live@gmail.com					

A portion of the List of selected participants for the Campaign:

From: AIASC Space <<u>aiasc@space-india.com</u>> Date: Tue, 13 Aug, 2024, 11:54 am Subject: AIASC 2024 PHASE 2 - CERTIFICATES To: <<u>snigdhakeshari123@gmail.com</u>>, <<u>mohitcbic@gmail.com</u>>

Dear Participants,

Greetings from the AIASC Team,

Congratulations on your participation in the All India Asteroid Search Campaign (AIASC) 2024, Phase 2.

Your participation certificate is attached to this mail.

Feel free to revert back over the same mail, in case of any query or change in the certificates.

Looking forward to your active participation in future events.

Happy Clear Skies!

Thanking You, **Team AIASC 2024 SPACE India** E: aiasc@space-india.com W: www.space-india.com

Email communication on successful identification of potential Asteroids

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All India Asteroid Search Campaign July 1 - 26, 2024									
	Provisional	# Object	Students	School	Location	Status	DateOfImage	Linked	
Total:	14	P21VUZY	Y. Soleimani	SPACE Titan	India	Preliminary	06/29/24	YSO2402	
		P21W6ui P21W4JX P21W3Sa P21W6r9 P21W6rH P21W6vg	N. Chauhan Z. Darrudi, N. Payami M. Mahapatra, D. Rajguru A. Pant, A. Sharda S. Keshari S. Mittal, D. Mantero	Jeevanbharti Atal Lab Team 1 SPACE Amazing Follower SPACE Antarix SPACE GEU 1 SPACE iAstronomer 21 SPACE Team 102	India India India India India India	Preliminary Preliminary Preliminary Preliminary Preliminary Preliminary	07/02/24 07/02/24 07/02/24 07/02/24 07/02/24 07/02/24	JBA0003 NPO2414 MAN0017 AAS0101 SNI2307 SHR1234	
		P21Whct P21Whbn P21Whbq P21Whbr P21Whbi P21WhUa P21WhUE	R.Adhikari, M.Solanki P. Kumar, A. Tripathi P. Kumar, A. Tripathi P. Kumar, A. Tripathi S. Memon, S. Tripathi A.Srikanes, L. Ram A.Srikanes, L. Ram	Euro School SPACE Champions SPACE Champions SPACE Champions SPACE Surat Team 1 SPACE Team 97 SPACE Team 97	India India India India India India India	Preliminary Preliminary Preliminary Preliminary Preliminary Preliminary	07/06/24 07/06/24 07/06/24 07/06/24 07/06/24 07/06/24	EUR0010 KDJ0003 KDJ0002 KDJ5149 SVT0010 SDMN013 VIK123	

Citizen Scientists who identified potential Asteroids during the Campaign

5011 (2022), 7,772
From: iascsearch <iascsearch@hsutx.edu></iascsearch@hsutx.edu>
Date: Thu, 19 Sept, 2024, 3:09 am
Subject: RE: AIASC 2024 PHASE 2 - CERTIFICATES
To: Snigdha Keshari < <u>snigdhakeshari123@gmail.com</u> >
Cc: <u>dkeshari@indianoil.in</u> < <u>dkeshari@indianoil.in</u> >
Hi Snigdha,
Thi Shiguna,
I have attached your report and the image set.
······································
The main steps for an asteroid are: 1) preliminary detection, 2) provisional, 3) numbered asteroid, 4)
named asteroid.
A preliminary detection is an original observation of an object. Not every preliminary detection
reaches provisional status. The object must be observed a second time within the next 7-10 days by a
sky survey. If it is, then the detection is changed to provisional status by the Minor Planet Center
(MPC).
Asteroid discoveries with provisional status are maintained in the MPC database for many years, until
there have been a sufficient number of observations to fully determine the orbit. This process can take
6-10 years.
When an asteroid has a fully determined orbit, it becomes numbered. At this point, the MPC assigns
an observatory with discovery credit. If the observatory credited is Pan-STARRS, and the object is one
that was found in an IASC campaign, IASC lets the team know that object has become numbered.
Objects that fit this criteria can then have a name suggested by the original team.
You can find out more about the PS2 telescope here: <u>https://about.ifa.hawaii.edu/facility/haleakala-</u>
observatories/#panstarrs
Thank you, and happy hunting!
Cassidy Davis

IASC Coordinator

Email communication with IASC on the current status of the discovered potential Asteroid.